FTR - Flight Test Report

Manufacturer		Type testing No.	EAPR-GS-0856/18	JE J J J
	AIRDESIGN GmbH Rhombergstraße 9 A-6967 Absam	serial number	Proto	Messen Prüfen Bewerten Rev. 2.3 - 26.11.2014
Model	Susi 3 - 13	Location	Brauneck	EAPR GmbH - Marktstr. 11 D-87730 Bad Grönenbach - Germany
Comment		Location	Achensee	

eise, vervielfältigt werden

Date of testing	16.07.2018	Minimum take o 50 kg	off weight	Maximum take off weight 85 kg		
Testpilot		Sepp Bauer		Mike Küng		
Harness		EAPR- Lightequipmen	t	Eapr-Testequipment		
Pilot's take off weig	nt	60	kg	87 kg		



est-criteria		Minimum take off weight	Evaluation	Maximum take off weight	Evaluation	
1. Inflation / take-off - 4.4.1						
Rising behavior		no pilot correction required	А	no pilot correction required	А	
Special take off technique required		No	A	No	A	
2. Landing - 4.4.2						
Special landing technique required		No	А	No	А	
3. Speeds in straight flight - 4.4.3						
Trim speed more than 30km/h		Yes	А	Yes	А	
Speed range using the controls larger than 10km/	′h	Yes	A	Yes		
Minimum speed		25 km/h to 30 km/h	В	25 km/h to 30 km/h		
4. Control movement - 4.4.4		23 KINI to 30 KINI		23 KIMI to 30 KIMI	В	
4. Control movement - 4.4.4			-			
Max. weight in flight up to 80kg			-		-	
Max. weight in flight 80 to 100kg		Increasing 45cm - 60cm	С	Increasing 45cm - 60cm	С	
Max. weight in flight greater than 100kg			-		-	
5. Pitch stability exiting accelerated flight - 4.	4.5	•		•		
Dive forward angle on exit		Dive forward less than 30°	А	Dive forward less than 30°	А	
Collapse occurs		No	A	No	A	
6. Pitch stability operating controls during acc	celerated	flight - 4.4.6				
Collapse occurs		No	А	No	А	
7. Roll stability and damping - 4.4.7					,,	
		Deducing		Detheine	А	
Oscillations		Reducing	A	Reducing	A	
8. Stability in gentle spirals - 4.4.8						
Tendency to return to straight flight		Spontaneous exit	A	Spontaneous exit	A	
9. Behaviour exiting a fully developed spiral d	live - 4.4.					
Initial response of glider (first 180°)		No immediate reaction	В	No immediate reaction	B	
Tendency to return to straight flight		Spontaneous exit	A	Spontaneous exit	A	
Turn angle to recover normal flight		720° to 1080°, spontaneous recovery B 1080° to 1440°, spontaneous recovery				
10. Symmetric front collapse - 4.4.10						
Folding lines used		No		No		
Entry	.30%	Rocking back less than 45°	A	Rocking back less than 45°	A	
Recovery	speed ~ 3(Spontaneous in less than 3 sec	А	Spontaneous in less than 3 sec	А	
Dive forward angle on exit	ds u	0° - 30° Keeping course	A	30° - 60° Entering a turn of less than 90°	В	
Cascade occurs	trim	No	A	No	A	
Entry	> 50%	Rocking back less than 45°	A	Rocking back less than 45°	A	
Recovery	5< beeds	Spontaneous in less than 3 sec	А	Spontaneous in less than 3 sec	А	
Dive forward angle on exit	- iu	0° - 30° Entering a turn of less than 90°	A	30° - 60° Entering a turn of less than 90°	В	
Cascade occurs		No	A	No	A	
Entry	50%	Rocking back less than 45°	A	Rocking back less than 45°	A	
Recovery	rated >	Spontaneous in less than 3 sec	А	Spontaneous in less than 3 sec	A	
Dive forward angle on exit	coeler	30° - 60° Keeping course	В	30° - 60° Entering a turn of less than 90°	В	
Cascade occurs	ac	No	A	No	A	
11. Exiting deep stall (parachutal stall) - 4.4.1	1					
Deep stall achieved		Yes		Yes		
Recovery		Spontaneous in less than 3 sec	А	-h		
Dive forward angle on exit		30° - 60°	В	30° - 60°	В	
Change of course		Changing course less than 45°	A	Changing course less than 45°	A	
Cascade occurs		No	A	No	А	

12. High angle of attack recovery - 4.4.12									
Recovery		Spontaneous in less than 3 sec			А	Spontaneous in less than 3 sec			А
Cascade occurs		No		A	No			A	
13. Recovery from a developed full stall - 4.4.	13								
Dive forward angle on exit Collapse		30° - 60° No collapse			B	60° - 90° No collapse			C A
Cascade occurs (other than collapse)		No Greater than 45			А	No			А
Rocking backward	Line tension		j°		C A	Less than 45° Most lines tight			A
14. Asymmetric collapse (trim speed) - 4.4.14		Most lines tight							
Folding lines used	0	No		1		No	•	1	
Change of course until re-inflation	ese	< 90°	Dive or roll angle	15° - 45°	А	< 90°	Dive or roll angle	15° - 45°	A
Re-inflation behavior	trim speed, max 50% collapse	Spontaneous re	-inflation		А	Spontaneous re	e-inflation		А
Total change of course	trim speed x 50% colla	Less than 360°		A	Less than 360° No No		A		
Collapse on the opposite side occurs Twist occurs	trii nax 5	No No		A			A		
Cascade occurs	2	No			Â	No			Â
Change of course until re-inflation	se	90° - 180°	Dive or roll angle	15° - 45°	В	< 90°	Dive or roll angle	60° - 90°	С
Re-inflation behavior	trim speed, max 75% collapse	Spontaneous re	e-inflation		А	Spontaneous re	e-inflation		А
Total change of course	i spe	Less than 360°			A	Less than 360°			A
Collapse on the opposite side occurs	trim ax 75	No		А	No			А	
Twist occurs Cascade occurs	Ê	No No		A	No No			A	
			Dia	150 150			Dia 1	450 000	
Change of course until re-inflation	d, tpse	90° - 180°	Dive or roll angle	15° - 45°	В	< 90°	Dive or roll angle	45° - 60°	С
Re-inflation behavior	accelerated, max 50% collapse	Spontaneous re	e-inflation		А	Spontaneous re	e-inflation		А
Total change of course	scele 50%	Less than 360°			A	Less than 360°			A
Collapse on the opposite side occurs Twist occurs	ac max :	No No			A	No No			A
Cascade occurs		No	1		A	No			A
Change of course until re-inflation	se	90° - 180°	Dive or roll angle	45° - 60°	С	90° - 180°	Dive or roll angle	60° - 90°	D
Re-inflation behavior	accelerated, max 75% collapse	Spontaneous re	-inflation		А	Spontaneous re	e-inflation		А
Total change of course	accelerated ix 75% colla	Less than 360°			A	Less than 360°			A
Collapse on the opposite side occurs	acc ax 7.	No			A	No			A
Twist occurs Cascade occurs	E	No No			A	No No			A
15. Directional control with a maintained asym	metric co	llapse - 4.4.15							
Able to keep course straight		Yes			A	Yes			A
180° turn away from the collapsed side possible in	n 10 sec	Yes		A	Yes			A	
Amount of control range between turn and stall or	spin	25% to 50% of the symmetric control travel		С	25% to 50% of the symmetric control travel			С	
16. Trim speed spin tendency - 4.4.16 Spin occurs		No				No			A
17. Low speed spin tendency - 4.4.17		INO			A	INO			A
Spin occurs		No			A	No			A
18. Recovery from a developed spin - 4.4.18		1							
Spin rotation angle after release		Stops spinning in less than 90°		А	Stops spinning in less than 90°			А	
Cascade occurs		No		A	No			A	
19. B-line-stall - 4.4.19 Change of course before release		Changing course	e less than 45°		A	Changing cours	e less than 45°		А
Behaviour before release		Changing course less than 45° Remains stable with straight span		A	Changing course less than 45° Remains stable with straight span			A	
Recovery		Spontaneous in less than 3 sec		A	Spontaneous in less than 3 sec			A	
Dive forward angle on exit Cascade occurs		0° - 30° No		A	30° - 60° No		A		
20. Big ears - 4.4.20									
Entry procedure	Standard technique			А	Standard technique			А	
Behaviour during big ears		Stable flight		A	Stable flight			A	
Recovery		Spontaneous in less than 3 sec		А	Spontaneous in less than 3 sec			А	
Dive forward angle on exit		0° - 30°		A	0° bis 30°			A	
21. Big Ears in accelerated flight - 4.4.21						-			
Entry procedure		Standard technique		А	Standard technique			А	
Behaviour during big ears		Stable flight		A	Unstable flight			С	
Recovery		Spontaneous in less than 3 sec		А	Spontaneous in less than 3 sec			А	
Dive forward angle on exit		0° - 30°		А	0° bis 30°			A	
Behaviour immediately after releasing the accelarator while maintaining big ears		Stable flight		А	Stable flight			А	
23. Alternative means of directional control - 4	4.4.22								
180° turn achievable in 20 sec		Yes		А	Yes			А	
Stall or spin occurs		No		00	А	No	-		A
23. Any other flight procedure and/or configur	ation des	cribed in the use	r's manual - 4.4.	23	NA				NA
Procedure works as described	Procedure works as described Procedure suitable for novice pilots								
					NA				NA
Procedure suitable for novice pilots Cascade occurs									NA NA
Procedure suitable for novice pilots					NA NA				