FTR - Flight Test Report

Manufacturer		Type testing No.	EAPR-GS-0848/18	JE J >>
	AIRDESIGN GmbH Rhombergstraße 9 A-6967 Absam	serial number	Proto	Messen Prüfen Bewerten Rev. 2.3 - 26.11.2014
Model	Susi 3 18	Location	Achensee	EAPR GmbH - Marktstr. 11 D-87730 Bad Grönenbach - Germany
Comment	testing with crossline	Location	Achensee	

eise, vervielfältigt werden

Date of testing	08.05.2018	Minimum take off 50 kg	weight	Maximum take off weight 70 kg		
Testpilot		Sepp Bauer		Mike Küng		
Harness		EAPR- Lightequipment	-6-	EAPR-Testequipment		
Pilot's take off weig	ht	60 kg		70 kg		

Classification	В



Test-criteria	Minimum take off weight	Evaluation	Maximum take off weight	Evaluation	
1. Inflation / take-off - 4.4.1					
Rising behavior	no pilot correction required	А	no pilot correction required	А	
Special take off technique required	No	A	No	A	
2. Landing - 4.4.2					
Special landing technique required	No	А	No	А	
3. Speeds in straight flight - 4.4.3	10	, A	10	~~~	
Trim speed more than 30km/h	Yes	A	Yes	А	
Speed range using the controls larger than 10km/h	Yes	A	Yes	A	
Minimum speed	Loop then OF hm/h	А	Less than 25 km/h		
4. Control movement - 4.4.4	Less than 25 km/h	A	Less than 25 killin	A	
4. Control movement - 4.4.4				_	
Max. weight in flight up to 80kg	Increasing > 55cm	А	Increasing > 55cm	A	
Max. weight in flight 80 to 100kg		-		-	
Max. weight in flight greater than 100kg		-		-	
5. Pitch stability exiting accelerated flight - 4.4.5	·				
Dive forward angle on exit	Dive forward less than 30°	A	Dive forward less than 30°	А	
Collapse occurs	No	A	No	A	
6. Pitch stability operating controls during accelera	ited flight - 4.4.6				
Collapse occurs	No	А	No	А	
7. Roll stability and damping - 4.4.7	10	N	10		
Oscillations	Reducing	A	Reducing	А	
	Reducing	A	Reducing	A	
8. Stability in gentle spirals - 4.4.8			0		
Tendency to return to straight flight	Spontaneous exit	A	Spontaneous exit	A	
9. Behaviour exiting a fully developed spiral dive -					
Initial response of glider (first 180°)	Immediate reduction of rate in turn			B	
Tendency to return to straight flight	Spontaneous exit	A	Spontaneous exit	A	
Turn angle to recover normal flight	Less than 720°, spontaneous recovery	A	720° to 1080°, spontaneous recovery	В	
10. Symmetric front collapse - 4.4.10					
Folding lines used	No		No		
Entry	Rocking back less than 45°	A	Rocking back less than 45°	A	
Deservery	Spontaneous in less than 3 sec	А	Spontaneous in less than 3 sec	А	
Biro ionala angio on ovit	F C CC Propriet	A	0° - 30° Entering a turn of less than 90°	A	
	10	A	No	A	
Entry	Rocking back less than 45°	A	Rocking back less than 45°	A	
	Spontaneous in less than 3 sec	A	Spontaneous in less than 3 sec	A	
Dive forward angle on exit	Reeping course	A	30° - 60° Entering a turn of less than 90°	В	
Cascade occurs	No	A	No	А	
Entry	Rocking back less than 45°	A	Rocking back less than 45°	A	
	Spontaneous in less than 3 sec	А	Spontaneous in less than 3 sec	А	
Dive forward angle on exit	30° - 60° Keeping course	В	30° - 60° Entering a turn of less than 90°	В	
Cascade Occurs	No	A	No	A	
11. Exiting deep stall (parachutal stall) - 4.4.11					
Deep stall achieved	Yes		Yes		
Recovery	Spontaneous in less than 3 sec	А	Spontaneous in less than 3 sec	А	
Dive forward angle on exit	30° - 60°	В	30° - 60°	В	
Change of course	Changing course less than 45°				
		No A No			

12. High angle of attack recovery - 4.4.12									
Recovery	Spontaneous in less than 3 sec			А	Spontaneous in	А			
Cascade occurs		No		A	No			A	
13. Recovery from a developed full stall - 4.4.	13								
Dive forward angle on exit Collapse		30° - 60° No collapse			B	30° - 60° No collapse			B A
Cascade occurs (other than collapse)		No			А	No			А
Rocking backward Line tension		Less than 45° Most lines tight			A	Less than 45° Most lines tight			A
14. Asymmetric collapse (trim speed) - 4.4.14		Wost mes ught				Most lines light			
Folding lines used		No				No			
Change of course until re-inflation	se	< 90°	Dive or roll angle	15° - 45°	А	< 90°	Dive or roll angle	15° - 45°	А
Re-inflation behavior	trim speed, max 50% collapse	Spontaneous re-inflation		A	Spontaneous re-inflation Less than 360°			A	
Total change of course	im sp 50%	Less than 360°		A				A	
Collapse on the opposite side occurs Twist occurs	nax	No No	No		A	No No			A
Cascade occurs	-	No			A	No			A
Change of course until re-inflation	esc	< 90°	Dive or roll angle	15° - 45°	А	90° - 180°	Dive or roll angle	15° - 45°	В
Re-inflation behavior	trim speed, max 75% collapse	Spontaneous re	e-inflation		А	Spontaneous re	-inflation		А
Total change of course Collapse on the opposite side occurs	rim s 75%	Less than 360° No No		A	Less than 360°			A	
Twist occurs	t max			A	No No			A	
Cascade occurs		No			А	No			A
Change of course until re-inflation	as	< 90°	Dive or roll angle	15° - 45°	А	90° - 180°	Dive or roll angle	15° - 45°	В
Re-inflation behavior	accelerated, max 50% collapse	Spontaneous re	-inflation		A	Spontaneous re	-inflation		А
Total change of course	cele 50%	Less than 360°			A	Less than 360°			A
Collapse on the opposite side occurs Twist occurs	ac 1ax {	No No			A	No No			A
Cascade occurs	e	No			A	No			A
Change of course until re-inflation	9	90° - 180°	Dive or roll angle	15° - 45°	В	90° - 180°	Dive or roll angle	15° - 45°	В
Re-inflation behavior	accelerated, max 75% collapse	Spontaneous re	-inflation	I	A	Spontaneous re	-inflation		A
Total change of course	celer 5%	Less than 360°			A	Less than 360°			A
Collapse on the opposite side occurs Twist occurs	ao lax 7	No No			A	No			A
Cascade occurs	E	No			A	No No			A
15. Directional control with a maintained asym	metric co	llapse - 4.4.15							
Able to keep course straight		Yes			A	Yes			A
180° turn away from the collapsed side possible in 10 sec		Yes		A	Yes				
Amount of control range between turn and stall or spin 16. Trim speed spin tendency - 4.4.16		More than 50% of the symmetric control travel		A	More than 50% of the symmetric control travel			A	
Spin occurs		No			A	No			A
17. Low speed spin tendency - 4.4.17									
Spin occurs		No			A	No			A
18. Recovery from a developed spin - 4.4.18		1				1			
Spin rotation angle after release Cascade occurs		Stops spinning in less than 90° No		A	Stops spinning in less than 90° No			A A	
19. B-line-stall - 4.4.19		110				110			
Change of course before release	Changing course less than 45°			A	Changing cours	A			
Behaviour before release		Remains stable with straight span		A	Remains stable with straight span			A	
Recovery		Spontaneous in less than 3 sec		А	Spontaneous in less than 3 sec			А	
Dive forward angle on exit		0° - 30°		A	30° - 60°			A	
Cascade occurs 20. Big ears - 4.4.20		No			A	No			A
-		Oberede al 1 and 1				Otennie of the star			
Entry procedure Behaviour during big ears		Standard technique Stable flight		A A	Standard technique Stable flight			A A	
Recovery		Spontaneous in less than 3 sec		A	Spontaneous in less than 3 sec			A	
Dive forward angle on exit		0° - 30°		А	0° bis 30°			А	
21. Big Ears in accelerated flight - 4.4.21									
Entry procedure		Standard technique		A	Standard technique			A	
Behaviour during big ears Becovery		Stable flight		A	Stable flight			A	
Recovery Dive forward angle on exit		Spontaneous in less than 3 sec 0° - 30°		A	Spontaneous in less than 3 sec 0° bis 30°			A	
Behaviour immediately after releasing the accelarator while maintaining big ears		Stable flight		A	Stable flight			A	
23. Alternative means of directional control	1.4.22	1							
180° turn achievable in 20 sec		Yes		А	Yes			A	
Stall or spin occurs	otion de l	No	via manual da	02	A	No			A
23. Any other flight procedure and/or configur Procedure works as descibed	auon des	under in the use	a sindhual - 4.4.	.23	NA				NA
Procedure suitable for novice pilots	<u>+</u>			NA				NA	
	Cascade occurs								NIA
Cascade occurs					NA				NA
		L							INA