FTR - Flight Test Report Dieser Prüfbericht darf ohne schriftliche Zustimmung der EAPR nicht, auch nich

Manufacturer	AIRDESIGN	Type testing No.	EAPR-GS-0770/18	
	AIRDESIGN GmbH Rhombergstraße 9 A-6967 Absam	serial number	XA11XXS1PP174329	
Model	Eazy 2 XXS	Leastion	Brauneck	
Comment	50kg testflight by sep. Testpilot	Location	Achensee	



Rev. 2.3 - 26.11.2014 EAPR GmbH - Marktstr. 11 D-87730 Bad Grönenbach - Germany

Date of testing	05.03.2018	Minimum take off w 50 kg	reight	Maximum take off weight 72 kg		
Testpilot		Sepp Bauer		Mike Küng		
Harness		EAPR- Lightequipment		EAPR-Testequipment		
Pilot's take off weigl	ht	50/65 kg		72 kg		

Classification



Test-criteria	st-criteria		finimum take off weight Evaluation Maximum take off weight			
1. Inflation / take-off - 4.4.1						
Rising behavior		no pilot correction required	Α	no pilot correction required	Α	
Special take off technique required		No	Α	No	Α	
2. Landing - 4.4.2		1.0	, , ,	1.0	, ,,	
Special landing technique required		No	l A	No	А	
		NO	A	140	A	
3. Speeds in straight flight - 4.4.3		Lv		Lv		
Trim speed more than 30km/h		Yes A Yes		Yes	Α	
Speed range using the controls larger than 10km/h		Yes	Α	Yes	Α	
Minimum speed		Less than 25 km/h	Α	Less than 25 km/h	А	
4. Control movement - 4.4.4						
Max. weight in flight up to 80kg		Increasing > 55cm	Α	Increasing > 55cm	Α	
Max. weight in flight 80 to 100kg			-		-	
Max. weight in flight greater than 100kg			-		-	
5. Pitch stability exiting accelerated flight - 4.	4.5	<u> </u>				
Dive forward angle on exit		Dive forward less than 30°	Α	Dive forward less than 30°	Α	
Collapse occurs			A	No	A	
6. Pitch stability operating controls during acc	elerated	flight - 4.4.6				
Collapse occurs		I No	Α	No	А	
7. Roll stability and damping - 4.4.7		1.0	, , ,	1.10	, ,,	
Oscillations		Reducing	l A	Reducing	l A	
		Reducing	A	heducing	A	
8. Stability in gentle spirals - 4.4.8						
Tendency to return to straight flight		Spontaneous exit	Α	Spontaneous exit	Α	
9. Behaviour exiting a fully developed spiral d	ive - 4.4.	Immediate reduction of rate in turn				
	ial response of glider (first 180°)		Α			
Tendency to return to straight flight		Spontaneous exit	A	Spontaneous exit	A	
Turn angle to recover normal flight		Less than 720°, spontaneous recovery	Α	Less than 720°, spontaneous recovery	Α	
10. Symmetric front collapse - 4.4.10						
Folding lines used		No		No		
Entry	~ 30%	Rocking back less than 45°	Α	Rocking back less than 45°	Α	
Recovery		Spontaneous in less than 3 sec	Α	Spontaneous in less than 3 sec	Α	
Dive forward angle on exit	paeds u	0° - 30° Keeping course	А	0° - 30° Keeping course	Α	
Cascade occurs	trim	No	Α	No	Α	
Entry	> 50%	Rocking back less than 45°	Α	Rocking back less than 45°	Α	
Recovery		Spontaneous in less than 3 sec	Α	Spontaneous in less than 3 sec	Α	
Dive forward angle on exit	paads u	0° - 30° Keeping course	Α	0° - 30° Entering a turn of less than 90°	Α	
Cascade occurs	Ē	No	A	No	A	
Entry	erated > 50%	Rocking back less than 45°	Α	Rocking back less than 45°	Α	
Recovery		Spontaneous in less than 3 sec	Α	Spontaneous in less than 3 sec	Α	
Dive forward angle on exit	celer	0° - 30° Keeping course	Α	0° - 30° Entering a turn of less than 90°	Α	
Cascade occurs	acce	No	Α	No	Α	
11. Exiting deep stall (parachutal stall) - 4.4.1	1					
Deep stall achieved		Yes		Yes		
Recovery		Spontaneous in less than 3 sec	Α	Spontaneous in less than 3 sec	Α	
		· ·		'		
Dive forward angle on exit		0° - 30°	A	0° - 30°	Α	
Change of course		Changing course less than 45°	A	Changing course less than 45°	A	
scade occurs		No	Α	No	Α	

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12. High angle of attack recovery - 4.4.12									
Recovery	Spontaneous in less than 3 sec		Α	Spontaneous in less than 3 sec			А		
Cascade occurs		No		Α	No			Α	
13. Recovery from a developed full stall - 4.4.13					•				
Dive forward angle on exit		0° - 30°			Α	0° - 30°			Α
Collapse Cascade occurs (other than collapse)		No collapse No			A	No collapse No			A
Rocking backward		Less than 45°			A	Less than 45°			A
Line tension		Most lines tight			Α	Most lines tight			А
14. Asymmetric collapse (trim speed) - 4.4.14		No				No			
Folding lines used		< 90°		00 150		< 90°	Dive or roll angle	150 450	^
Change of course until re-inflation	bse	< 90-	Dive or roll angle	0° - 15°	Α	< 90°	Dive or roll angle	15° - 45°	Α
Re-inflation behavior	trim speed, max 50% collapse	Spontaneous re-i	nflation		Α	Spontaneous re	-inflation		Α
Total change of course	trim speed x 50% colla	Less than 360°			Α	Less than 360°			Α
Collapse on the opposite side occurs Twist occurs	nax tri	No No		A	No No			A	
Cascade occurs	_	No			A	No			A
Change of course until re-inflation	Ф	< 90°	Dive or roll angle	15° - 45°	Α	< 90°	Dive or roll angle	15° - 45°	Α
De tefferte behante.	trim speed, max 75% collapse	0	- n - r	1	•	0	1-0-0	l	
Re-inflation behavior	obee ocol	Spontaneous re-inflation			Α	Spontaneous re		Α	
Total change of course Collapse on the opposite side occurs	trim speed x 75% colls	Less than 360° No			A	Less than 360° No			A
Twist occurs	mag.	No			Α	No			Α
Cascade occurs		No		Α	No			Α	
Change of course until re-inflation	_	< 90°	Dive or roll angle	0° - 15°	Α	< 90°	Dive or roll angle	15° - 45°	А
	d, apse			L			<u> </u>	L	
Re-inflation behavior	accelerated, max 50% collapse	Spontaneous re-i	nflation		Α	Spontaneous re	-inflation		Α
Total change of course	ccele 50%	Less than 360°			A	Less than 360°			A
Collapse on the opposite side occurs Twist occurs	max	No No			A	No No			A
Cascade occurs		No			A	No			A
Change of course until re-inflation	Φ	< 90°	Dive or roll angle	15° - 45°	Α	< 90°	Dive or roll angle	15° - 45°	Α
Re-inflation behavior	accelerated, max 75% collapse	Spontaneous re-i	inflation	1	Α	Caantanaaya ra	inflation	1	Α
	lerat % co		Tillation			Spontaneous re-inflation			
Total change of course Collapse on the opposite side occurs	ассе x 75°	Less than 360° No No			A	Less than 360° No	A		
Twist occurs	ma				Α	No			Α
Cascade occurs		No A 4 1 5			Α	No			Α
15. Directional control with a maintained asymptotic to keep course straight	metric co	Yes			Α	Yes			Α
180° turn away from the collapsed side possible in	n 10 coc	Yes			A	Yes			A
Too turn away from the collapsed side possible in 10 sec						103			
Amount of control range between turn and stall or spin		More than 50% of the symmetric control travel			Α	More than 50%	Α		
16. Trim speed spin tendency - 4.4.16									
Spin occurs	pin occurs		No			No	Α		
17. Low speed spin tendency - 4.4.17		I No.				No			
Spin occurs 18. Recovery from a developed spin - 4.4.18		No			Α	INO			Α
						I			
Spin rotation angle after release		Stops spinning in less than 90°			Α	Stops spinning in less than 90°			Α
Cascade occurs		No			Α	No	Α		
19. B-line-stall - 4.4.19 Change of course before release		Changing course less than 45°			Α	Changing cours	Α		
Behaviour before release				A	Remains stable with straight span			A	
23		Remains stable with straight span			^				^
Recovery		Spontaneous in less than 3 sec			Α	Spontaneous in less than 3 sec			Α
Dive forward angle on exit		0° - 30°			A	30° - 60°	A		
Cascade occurs 20. Big ears - 4.4.20		No			Α	No			Α
		Oranda III II				0			
Entry procedure		Standard technique			A	Standard technique			A
Behaviour during big ears		Stable flight			A	Stable flight			A
Recovery		Spontaneous in less than 3 sec			Α	Spontaneous in less than 3 sec			Α
Dive forward angle on exit		0° - 30°			Α	0° bis 30°			А
21. Big Ears in accelerated flight - 4.4.21									
Entry procedure		Standard technique		Α	Standard technique			Α	
Behaviour during big ears		Stable flight		Α	Stable flight			Α	
Behaviour during big ears	Recovery		Spontaneous in less than 3 sec		Α	Spontaneous in less than 3 sec			Α
		0° - 30°		Α	0° bis 30°			Α	
Recovery Dive forward angle on exit		0° - 30°				Stable flight			Α
Recovery Dive forward angle on exit Behaviour immediately after releasing the accelar	ator while	0° - 30° Stable flight			Α	Stable flight			/ \
Recovery Dive forward angle on exit					А	Stable flight			,,,
Recovery Dive forward angle on exit Behaviour immediately after releasing the accelar maintaining big ears 23. Alternative means of directional control -		Stable flight							
Recovery Dive forward angle on exit Behaviour immediately after releasing the accelar maintaining big ears 23. Alternative means of directional control - 180° turn achievable in 20 sec		Stable flight Yes			A	Yes			А
Recovery Dive forward angle on exit Behaviour immediately after releasing the accelar maintaining big ears 23. Alternative means of directional control - 180° turn achievable in 20 sec Stall or spin occurs	4.4.22	Stable flight Yes No	s manual - 4 4 4	23					
Recovery Dive forward angle on exit Behaviour immediately after releasing the accelar maintaining big ears 23. Alternative means of directional control - 180° turn achievable in 20 sec	4.4.22	Stable flight Yes No	s manual - 4.4.	23	A	Yes			А
Recovery Dive forward angle on exit Behaviour immediately after releasing the accelar maintaining big ears 23. Alternative means of directional control - 180° turn achievable in 20 sec Stall or spin occurs 23. Any other flight procedure and/or configur Procedure works as descibed Procedure suitable for novice pilots	4.4.22	Stable flight Yes No	s manual - 4.4.	23	A A NA NA	Yes			A A NA NA
Recovery Dive forward angle on exit Behaviour immediately after releasing the accelar maintaining big ears 23. Alternative means of directional control - 180° turn achievable in 20 sec Stall or spin occurs 23. Any other flight procedure and/or configure Procedure works as descibed	4.4.22	Stable flight Yes No	s manual - 4.4.	23	A A	Yes			A A

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